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HELICON PLASMA INJECTOR AND ION CYCLOTRON ACCELERATION DEVELOPMENT IN THE VASIMR EXPERIMENT

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ABSTRACT

In the Variable Specific Impulse Magnetoplasma Rocket (VASIMR) radio frequency (rf) waves both produce the plasma and then accelerate the ions. The plasma production is done by action of helicon waves. These waves are circular polarized waves in the direction of the electron gyromotion. The ion acceleration is performed by ion cyclotron resonant frequency (ICRF) acceleration. The Advanced Space Propulsion Laboratory (ASPL) is actively developing efficient helicon plasma production and ICRF acceleration. The VASIMR experimental device at the ASPL is called VX-10. It is configured to demonstrate the plasma production and acceleration at the 10kW level to support a space flight demonstration design. The VX-10 consists of three electromagnets integrated into a vacuum chamber that produce magnetic fields up to 0.5 Tesla. Magnetic field shaping is achieved by independent magnet current control and placement of the magnets. We have generated both helium and hydrogen high density ($>10^{18} \text{ m}^{-3}$) discharges with the helicon source. ICRF experiments are underway. This paper describes the VX-10 device, presents recent results and discusses future plans.

in helicon¹ devices, where a right-hand circularly polarized wave is launched into the plasma. Also in fusion research, ion cyclotron resonant heating (ICRH)² has been very successful at producing energetic ions without having electrodes or grids in contact with the intense plasma. The same concept of rf production and heating of magnetized plasmas can be used for high performance propulsion.

Efficient helicon plasma production and injection into a strong magnetic field is crucial for the performance of the Variable Specific Impulse Magnetoplasma Rocket (VASIMR) concept.^{3, 4} The subsequent ICRF acceleration of the ions via rf driven waves is necessary to produce significant thrust. The experiment in the Advanced Space Propulsion Laboratory (ASPL) at the NASA Johnson Space Center is presently focused on research in these two areas. The experimental device in the ASPL is called VX-10. It is configured to operate in the 10 kW power range to support the design of an early space flight demonstration of the VASIMR concept.⁵ This paper reports on the status of the helicon plasma source development and ICRF acceleration research.

INTRODUCTION

High power density plasma propulsion would vastly improve human exploration of our solar system. The fusion energy research community has long known that magnetized plasma systems enable high energy density discharges, and that the magnetic field insulates the material wall from the hot plasma. Radio frequency (rf) waves have proven to be efficient at plasma production

THE VX-10 EXPERIMENT

The VX-10 experiment is quickly evolving to a configuration that demonstrates the plasma production and ion acceleration components of the VASIMR concept. The VX-10 will soon operate at a total power level of 10 kW and targets a specific impulse of 5,000 to 10,000 seconds. Figure 1 shows two main configurations in the recent experimental evolution. The primary focus of the work so far has been helicon

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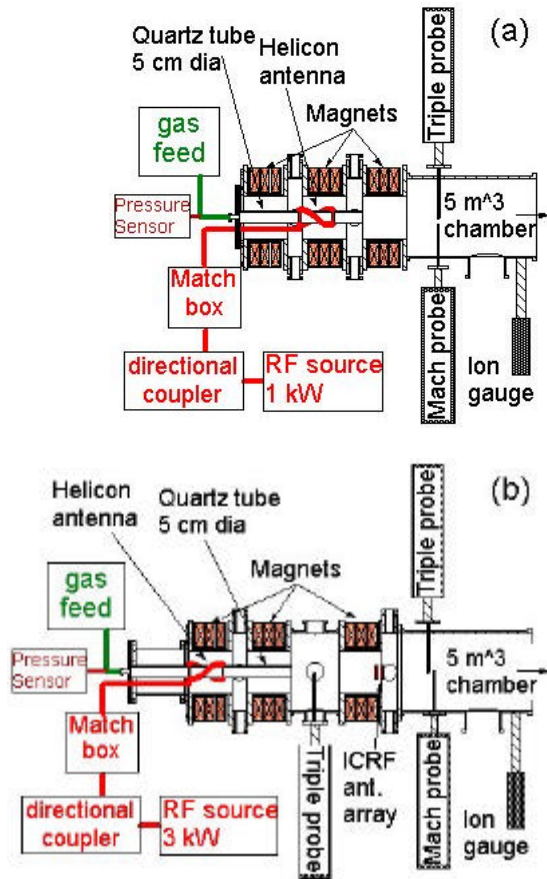


FIG. 1. Experiment configurations. (a) Helium helicon study configuration. (b) Low field helicon configuration with magnetic choke for hydrogen experiments.

plasma production with helium and hydrogen. The earlier configuration (Fig. 1a) was designed to produce an almost uniform magnetic field for just helicon study. The later configuration incorporates what we call a low field helicon section and magnetic choke. This configuration also has shifted the helicon antenna location and moved the gas injection plate to a much lower magnetic field location. The ICRF antenna is installed near the magnetic choke. Figure 2 shows the VX-10 in operation.

In the present configuration (Fig. 1b), neutral gas (hydrogen or helium in the present data) is injected at less than or about 1 mg/s into an axisymmetric magnetic field that is shaped in strength along the axis. The injected neutral gas is contained in a quartz tube (5 cm in diameter) until it is ionized by an rf driven helicon wave in a magnetic field on the order of 0.1 Tesla. The plasma then flows near the sound speed, about 10^4 m/s, into a high magnetic field, about 0.2 Tesla for hydrogen, region where ICRF acceleration takes place. The accelerated ions (~100 eV) move out

in an expanding magnetic nozzle where perpendicular velocity is converted to axial velocity.

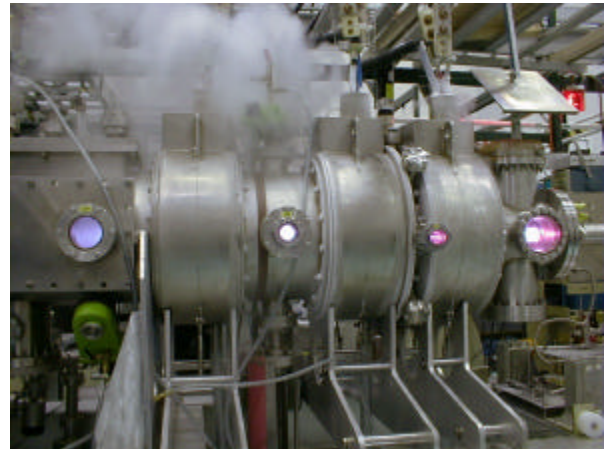


FIG. 2. The VX-10 device running.

The magnetic field in the experiment is produced using three independently driven liquid nitrogen cooled copper coil electromagnets capable of a maximum field of 0.3 Tesla for 10 minutes. A one Tesla capability for short pulses, 30 seconds, is being implemented. Light weight high temperature superconducting magnets are being developed to replace the low cost copper magnets.⁶ The VX-10 will soon start evolving to a superconducting magnet system.

A precision mass flow controller (MFC) measures and regulates gas flow into the system. The MFC range is 10 – 1000 standard cubic centimeters per minute (sccm). This corresponds to maximum mass flow of about 1.5 mg/second for hydrogen. The MFC also has 100 ms time response for gas pulsing experiments. A baratron measures the pressure in the gas inlet. A magnetically shielded ion gauge measures the neutral pressure in the vacuum chamber at the plasma exhaust region. The plasma exhausts into a 5 m³ vacuum chamber, which is pumped at 5000 liters/second. Components for a 50,000 liter/second pump upgrade have been acquired and installation is being planned.

In the helicon section, a water-cooled half-turn helical antenna is driven at 7 – 50 MHz with up to 3 kW of power. Figure 3 shows a photograph of an antenna. The 50 Ohm transmission line is matched to the antenna impedance using two vacuum variable capacitors in a simple network. Forward and reflected powers are measured with a directional coupler in the impedance matched section on the transmitter side of the matching network.

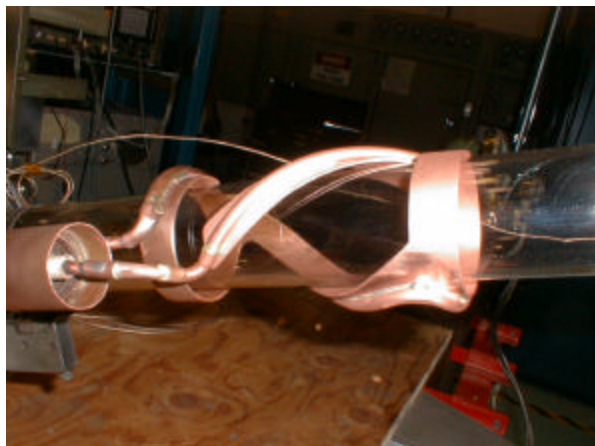


FIG. 3. Helicon antenna with the quartz tube running through it.

A water-cooled phased double half-turn antenna array is installed to launch waves at about 3 MHz for ICRF acceleration of the ions. The antenna array is installed just downstream of the magnetic choke and just upstream from the ion resonance. The two antenna rings are separated about 0.03 m axially and the rf power feeds are rotated 90°. Figure 4 shows the antenna array being installed.

A 100 kW rf steady state source is in operation at 3 MHz to drive the antenna array, although the array is driven only at the 10 kW power level. The power splitting, phasing, and impedance matching networks have been installed and tested.

DIAGNOSTICS

Reciprocating Langmuir and Mach probes are the primary plasma diagnostics. The Langmuir probe measures electron density and temperature profiles while the Mach probe measures flow profiles. Together this gives total plasma particle flux. The Langmuir probe has four molybdenum tips that are biased as a triple probe⁷, with an extra tip for measuring electrostatic fluctuations. The Mach probe has two molybdenum tips biased in ion saturation, one upstream and one downstream of a stainless steel separator.⁸

Retarding potential analyzer (RPA) diagnostics are being installed to measure the accelerated ions. A 70 GHz microwave density interferometer will soon be installed in the plasma exhaust region. It will be located about 0.3 m downstream of the ICRF resonance. There is also a CCD camera installed in the vacuum chamber that has an oblique view of the antennas and discharge.

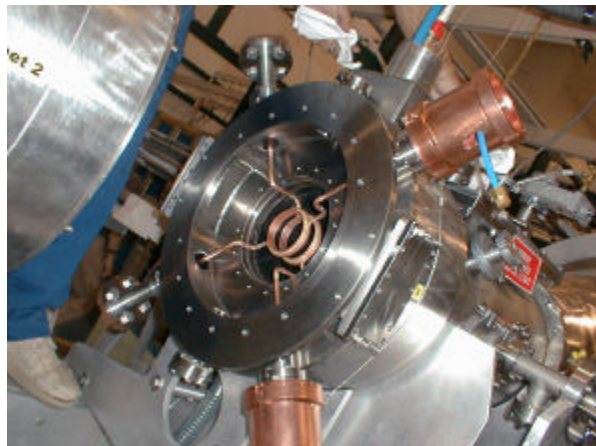


FIG. 4. The ICRF antenna array being installed.

Accounting for all of the power losses in the system is very important, so we have added several diagnostics to measure these quantities. Water calorimeters measure heat loads on all of the antennas. Thermocouples are used to measure heat loads on components such as the quartz tube. A bolometer has been recently installed to measure the total radiated power. To complement these measurements, we are installing rf antenna current and voltage diagnostics to determine the plasma loaded power from the antennas. This is an active area of study, so results will be presented in subsequent papers.

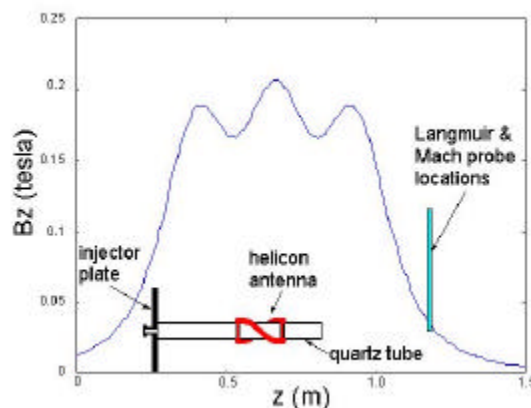


FIG. 5 A typical axial magnetic field profile corresponding to Fig. 1a. The antenna configuration and Langmuir probe locations are shown.

HELICON RESULTS

Configuration variations of the helicon section are being explored to optimize plasma production efficiency. Both hydrogen and helium discharges studies have been performed. Many parameters are

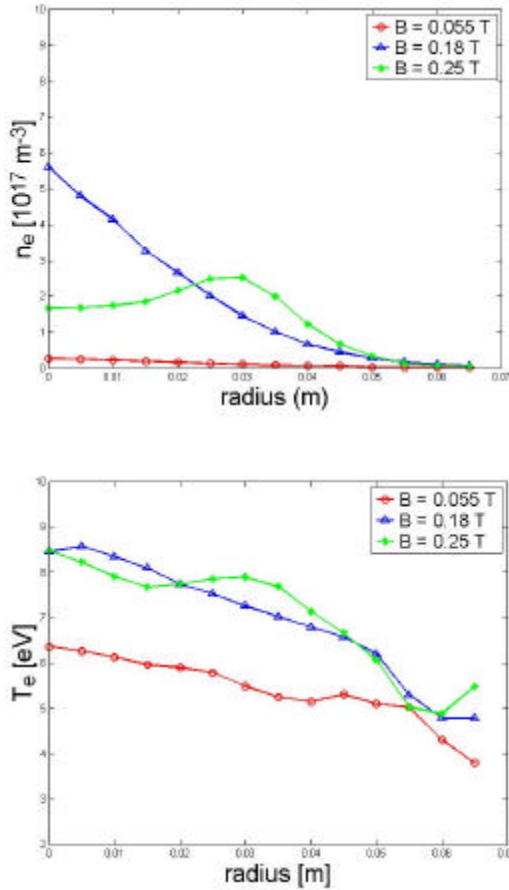


FIG. 6 Radial profiles for 600 W at 7 MHz with helium. (a) The electron density profile. (b) The electron temperature profile.

varied, such as: rf drive frequency and power; magnetic field shape, strength and polarization; quartz tube length and diameter; antenna structure and location; and gas flow rates.

HELIUM

Parametric studies presented here start with helium discharges in a fairly uniform magnetic field, and no magnetic choke. A typical magnetic field axial profile is shown in Fig. 5. In this study, the helicon antenna is nearly centered in the three-magnet set, shown in Fig. 1a. The quartz tube containing neutral gas extends through the antenna and a short distance beyond. The three magnets were configured in series and spaced equally.

In this configuration, we scanned the gas flow, magnetic field and input power for three rf drive frequencies. Figure 6 shows typical radial profiles of density and temperature for a 7 MHz driven case.

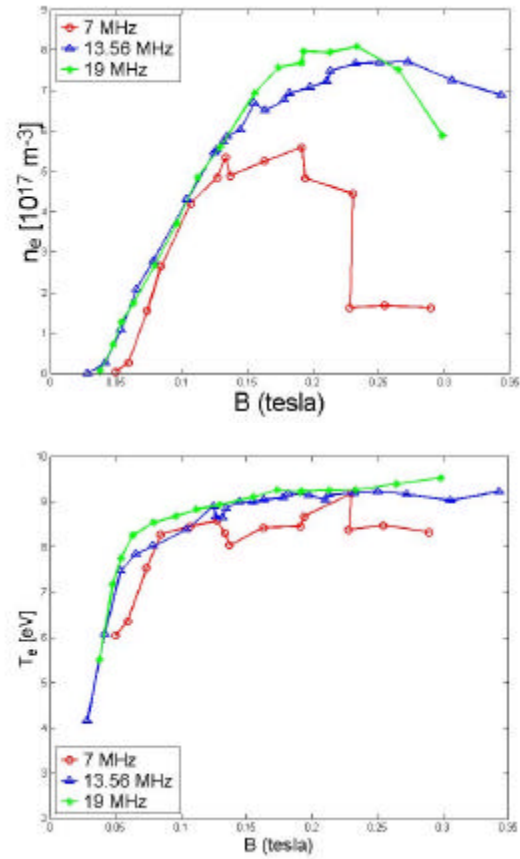


FIG. 7. Magnetic field scan for three rf frequencies. (a) The central electron density. (b) The central electron temperature.

(There is some small uncertainty to the plasma center, but the $r = 0$ point on the graph is within 1 cm of the center.) The density profiles are usually peaked and the temperature profiles are flat. We did see a mode change to a hollow profile at higher magnetic fields with the 7 MHz drive frequency. The 0.18T profiles shown in Fig. 6 have a typical shape of the other rf drive frequency cases.

In Fig. 7 we see the effect of scanning only the magnetic field strength. Power is fixed at 600 W and gas flow at 200 sccm ($9 \times 10^{19} / \text{sec}$) of helium. The plasma density rise starts at or about when the rf frequency equals the lower hybrid frequency (ω_{LH}), as defined by the following equation⁹

$$\frac{1}{\omega_{LH}^2} = \frac{1}{\omega_{ce} \omega_{ci}} + \frac{1}{\omega_{pi}^2 + \omega_{ci}^2}. \quad (1)$$

The density peaks when $\omega_f = 0.25\omega_{LH}$. The density quickly falls off at higher magnetic fields. The electron temperature, shown in Fig. 7b, rises quickly at the first

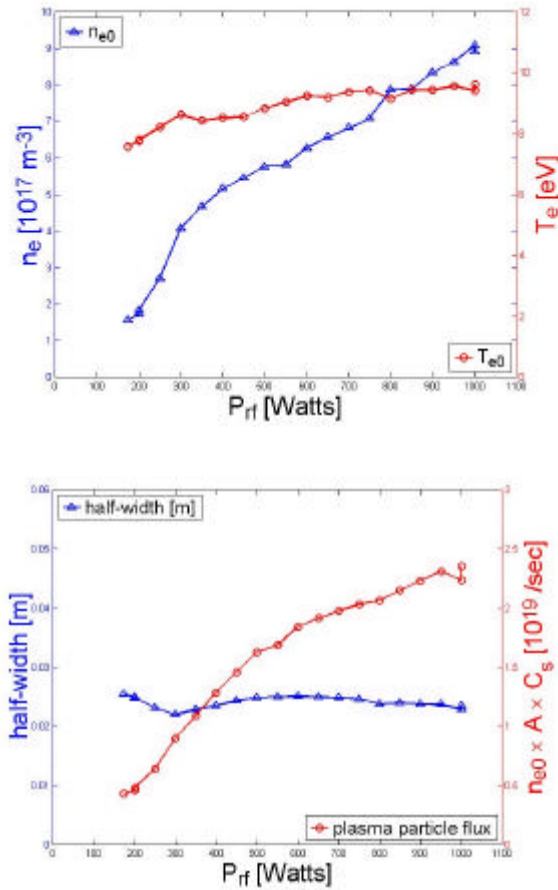


FIG. 8 Power scan at 13.56 MHz with 200 sccm of helium. (a) Central electron density and temperature. (b) Profile half-width and plasma flux estimate.

density rise, but then changes very little. The higher frequency generates a slightly higher temperature. These results indicate that higher frequency operation could lead to better plasma production performance.

Figure 8 shows the result of scanning the rf power for a 13.56 MHz drive frequency with 200 sccm mass flow of helium. Below about 150 W, a high density helicon discharge can not be generated, but a very low density discharge is still seen. After the helicon discharge is well started, above 300 W, the density rises almost linearly with the applied power. The electron temperature rises very slowly and stays about 7 eV. The profile shape almost stays constant as shown by the half-width plotted in Fig. 8b.

Also shown in Fig. 8b is an estimate of the total plasma flux (Γ_{plasma}) in the exhaust. This is estimated by

$$\Gamma_{plasma} \approx n_{e0} \cdot A \cdot C_s, \quad (2)$$

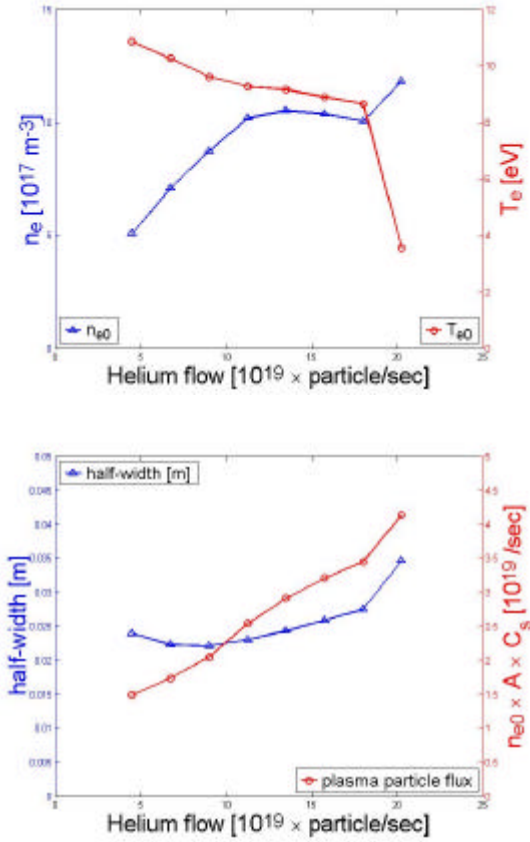


FIG. 9 Injected gas flow scan for 1000 W at 13.56 MHz. (a) Central electron density and temperature. (b) Profile half-width and plasma flux estimate.

where n_{e0} is the central electron density, A is the effective cross sectional area of the plasma column as determined by the Langmuir probe profile, and $C_s = \sqrt{T_e / m_i}$ is the plasma sound speed. This formula assumes a sonic flow speed of the plasma, since the Mach probe typically measures a sonic ($Ma \gg 1$) flow speed. The plasma flux estimate almost rises proportional to the applied power. At the maximum power, about 25% of the injected neutral gas gets ionized.

Figure 9 plots the effect of scanning the injected gas rate for fixed power at 1000 W. Below flow rates of about $5 \times 10^{19} / \text{sec}$, high density discharges could not be generated. Above flow rates of about $20 \times 10^{19} / \text{sec}$, the background vacuum chamber neutral pressure is very high and may be effecting the downstream measurements. Between the flow rate limits, the plasma density and flux estimate generally

rise with input gas flow rate. The electron temperature trends downward and the profiles tend to broaden. About 25% of the injected gas becomes ionized, independent of the input flow rate.

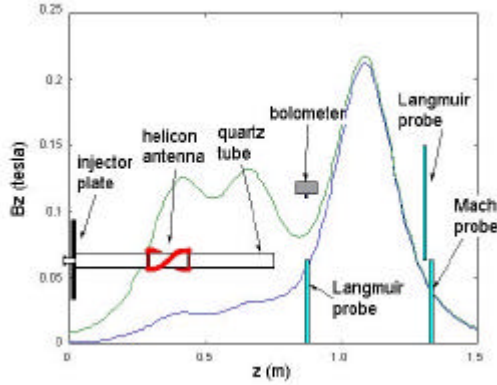


FIG. 10. Two typical magnetic field axial profiles corresponding to Fig. 1b. The helicon antenna configuration and Langmuir probe locations are shown.

HYDROGEN

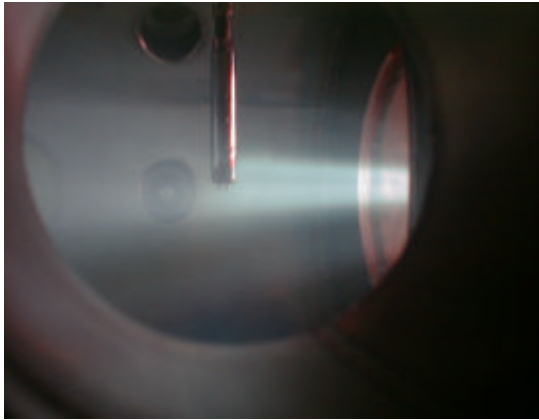


FIG. 11. Hydrogen plasma exhaust with the Langmuir probe inserted.

Hydrogen helicon discharge studies performed at the Oak Ridge National Laboratory¹⁰ discovered that high density plasma production is only achieved near the lower hybrid frequency in the VX-10 type of configuration. We reconfigured the VX-10 device, as figure 1b shows, in preparation of hydrogen experiments with a lower hybrid resonance in the helicon section. Figure 10 shows the typical magnetic profile. The plasma exhaust with the Langmuir probe inserted is shown in Fig. 11.

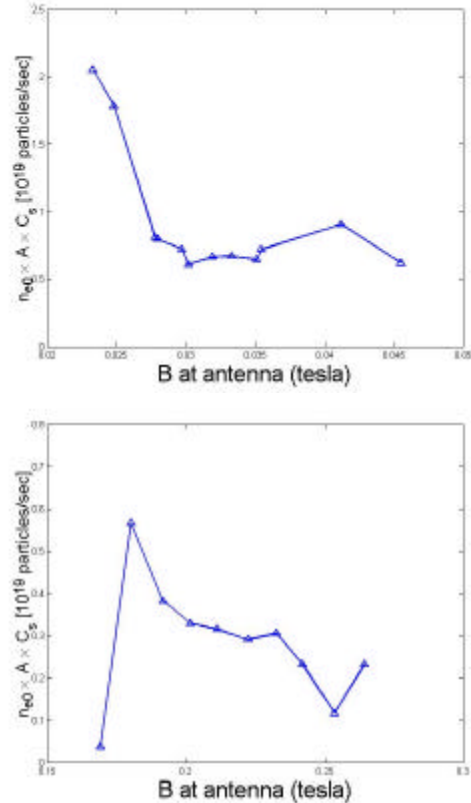


FIG. 12. Plasma flux estimate measure during magnetic field scans for hydrogen discharges with 2000 W at 25 MHz. (a) The field under the helicon antenna scan with the choke field fixed. (b) The choke field scanned with the field at the antenna fixed.

The VX-10 also sees the close connection to the lower hybrid frequency. We see a very strong density dependence on the magnetic field when $\omega_f \approx \omega_{LH}$ as Fig. 12a shows. Here we carefully scanned the magnetic field in the low field region near the antenna. The hydrogen flow was 70 sccm and the input rf power was 2 kW at 25 MHz. We see a strong peak in the density at a specific field strength. This field is somewhat lower than that required for lower hybrid resonance because the field profile is not flat and the resonance may occur downstream at a higher magnetic field. High density discharges are only achieved when the magnetic polarization is such that the right-hand wave is launched downstream.

The experiments also find that a high magnet field choke is required to achieve high density discharges. Figure 12b shows a sharp cutoff in density at a magnetic choke field of about 0.18 T. This corresponds to a magnetic field ratio from the choke to antenna of over 5. This indicates that some axial confinement may be required.

The hydrogen helicon experiments are fairly recent. Exploration with these type discharges are continuing. We are already seeing nearly 30% of the incoming gas ionized. The densities are on the order of what is necessary for a plasma thruster device.

FUTURE PLANS

Intense ICRF studies are beginning. Concomitantly, hydrogen helicon source optimization will continue. Presently, we are installing a high power (10 kW) vacuum antenna feed for the helicon system. The ICRF system is nearly complete and experiments will begin with this system in June of this year. A substantial helicon source upgrade will be performed by August that will allow nearly steady state high power operation with both the helicon and ICRF. This upgrade will include a larger diameter helicon source and a rf baffle to prevent power deposition of the ICRF on the upstream ion resonance in the source.

CONCLUSION

The Advanced Space Propulsion Laboratory recently configured a device called VX-10. This device focuses on the development of helicon plasma production and ion cyclotron resonant frequency acceleration to support an early space flight demonstration of the VASIMR concept at a 10 kW power level. We have described the VX-10 system with several of its diagnostic subsystems. High density helium and hydrogen helicon plasma discharges have been generated. We presented parametric (rf power, magnetic field and injected gas flow rate) scans in helium discharges for 3 fixed rf frequencies (7, 13.56 and 19 MHz). The fraction of injected neutral gas that gets ionized increase linearly with rf power and was measured as 25%, at maximum power. The data indicate that higher rf frequency may lead to better efficiency.

Hydrogen discharges with densities in excess of $1 \times 10^{18} / m^3$ were generated with the rf frequency (25 MHz) near the lower hybrid frequency. Strong dependence on the magnetic field was observed. Also, a downstream magnetic field with a choke ratio of about 5 seems necessary for high density operation. Approximately 30% of the injected gas was ionized. Ion cyclotron resonant frequency systems have been installed and tested. Experiments in this area are beginning in the next few months.

NOMENCLATURE

A Plasma cross sectional area.

C_s	Plasma sound speed.
M_a	Mach number.
m_i	ion mass.
n_{e0}	Central electron density.
T_e	Electron temperature.
sccm	Standard cubic centimeters per minute.
Γ_{plasma}	Plasma flux estimate.
ω_{ce}	electron cyclotron frequency.
ω_{ci}	ion cyclotron frequency.
ω_{LH}	lower hybrid frequency.
ω_{pi}	ion plasma frequency.
ω_{rf}	rf drive frequency.

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